

# Visibility study of Galileo satellites from a VLBI network

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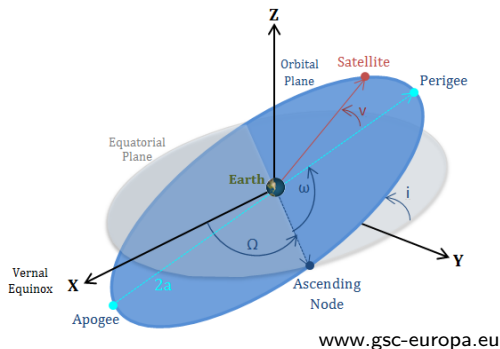
## Satellite observations with VLBI

- observations to quasars are integral part of VLBI
- VLBI transmitter on satellites would enable observing satellites with VLBI telescopes
- observations to satellites
  - enable interesting scientific applications
  - place new requirements

→ newly developed satellite scheduling module in VieSched++

# Satellite observations with VLBI Applications

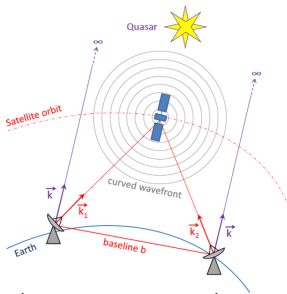
- geodetic space ties for connecting reference frames
- improvement of geodetic products, in terms of ITRF
- determination of absolute orientation of satellite constellation → Right Ascension of Ascending Node ( $\Omega$ )



# Satellite observations with VLBI

## Requirements

- cross-eyed observation geometry
- satellite is moving while observing → antenna has to follow
- conditions have to be checked continuously during a scan
  - visibility (horizon mask)
  - antenna slew rates
  - sun distance
  - cable wrap

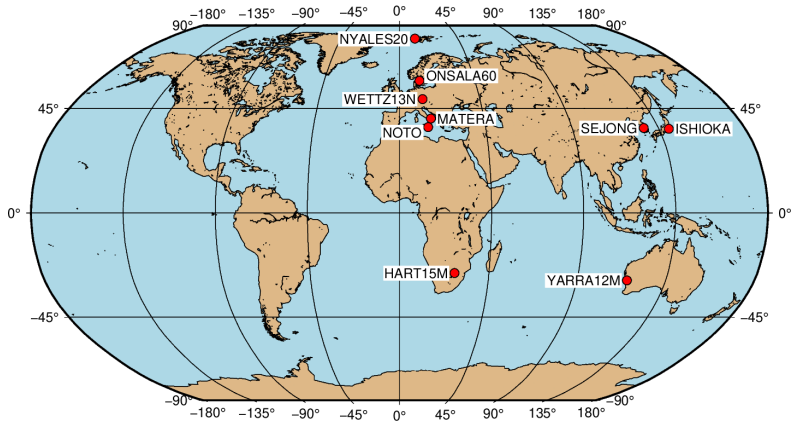


(Hellerschmied, 2014)

## Visibility study of Galileo satellites

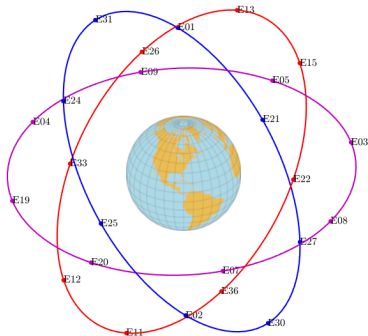
- 24 - hour IVS - R1978 session
- December 21, 2020 00:00 UTC
- 9 station network
- satellite scheduling module of VieSched++ was used to determine all possible observations to Galileo satellites
- observations are evaluated through
  - visibility (number of stations)
  - observation geometry

# VLBI network



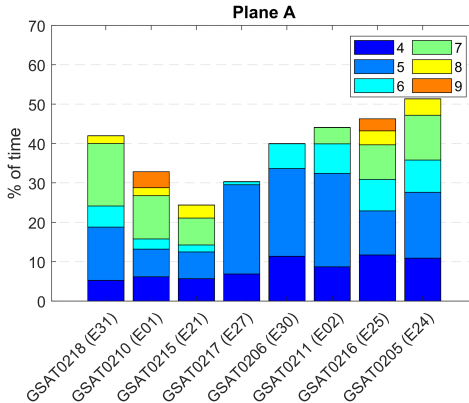
# Galileo satellite system

- 24 satellites<sup>1</sup>
- Medium Earth Orbit (MEO) in 23.222 km altitude
- three orbital planes (A, B and C) → spaced by 120°
- inclination of 56°
- eight satellites in each plane  
→ angular distance of 45°
- ground track repeat period of 10 days



<sup>1</sup>considering the current number of active satellites

# Visibility in percent

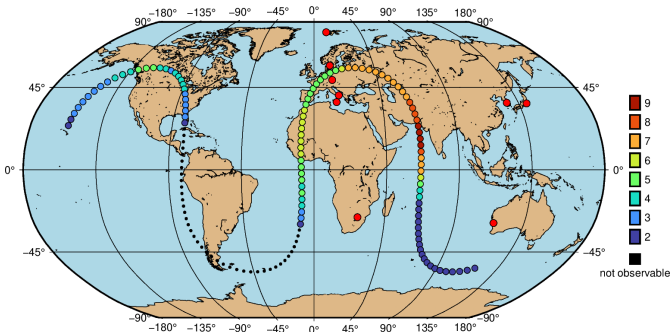


Percentage of time the satellites in Plane A, ordered by their slot number, are observable during the 24-hour IVS-R1978 session on December 21, 2020, color-coded by the number of stations from which the satellite is observable. A cut-off elevation angle of five degrees is considered.



## Visibility GSAT0216 (E25)

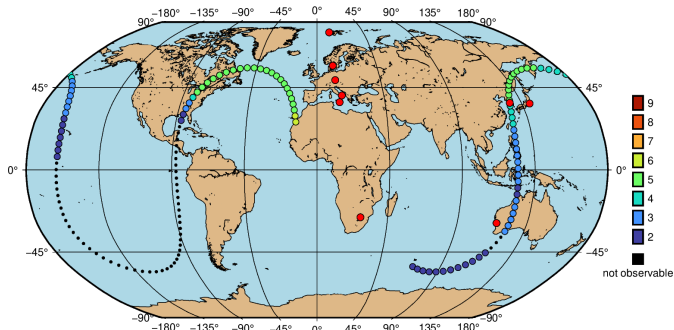
- rather high visibility due to course of ground track



Ground track of GSAT0216 (E25) during the 24-hour IVS-R1978 session on December 21, 2020. The dots represent the position of the satellite over Earth in a ten minute interval color-coded by the number of stations from which the satellite is visible. The smaller black dots represent positions where the satellite is not visible from at least two stations.

## Visibility GSAT0217 (E27)

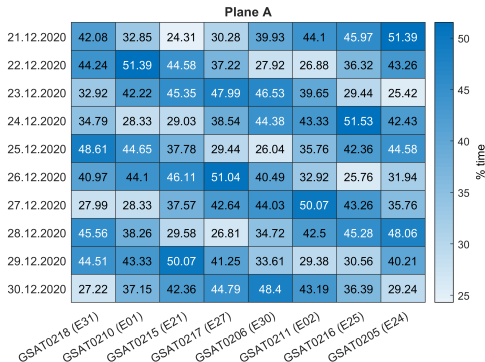
- rather low visibility due to course of ground track



Ground track of GSAT0217 (E27) during the 24-hour IVS-R1978 session on December 21, 2020. The dots represent the position of the satellite over Earth in a ten minute interval color-coded by the number of stations from which the satellite is visible. The smaller black dots represent positions where the satellite is not visible from at least two stations.

# Visibility over repeat period

- visibility of satellites is changing from day to day



Percentage of time the satellites in Plane A, ordered by their slot number, are observable from at least four stations over a period of 10 days starting on December 21, 2020. A cut-off elevation angle of five degrees is considered.

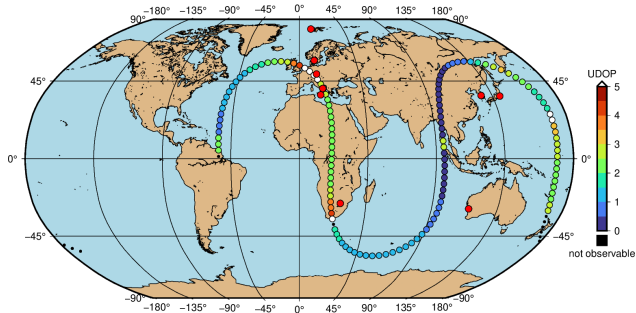
## Observation geometry

- precision of UT1-UTC depends on observation geometry
- UT1-UTC Dilution of Precision factor as indicator for sensitivity of satellite observation to UT1-UTC
- low UDOP factor means
  - better observation geometry
  - observation is more sensitive to UT1-UTC

→ see Appendix

## Observation geometry GSAT0102 (E12)

- low UDOP, if visible for stations forming E-W baselines
- high UDOP, if visible for stations forming N-S baselines



Ground track of GSAT0102 (E12) during the 24-hour IVS-R1978 session on December 21, 2020. The dots represent the position of the satellite over Earth in a ten minute interval color-coded by the values of the UDOP factor. The smaller black dots represent positions where the satellite is not visible from at least two stations.

## Summary

- satellite observations provide new possibilities
- no difference of visibility of the
  - three orbital planes
  - satellites within the plane
- visibility and observation geometry depends on
  - distribution of observing stations
  - session time
- Galileo exhibit high visibility

## References

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- Hellerschmied, A. (2014). *Observations of GNSS satellites with VLBI antennas - From observation planning to practical implementation*, Master thesis, Technical University of Vienna.
- Wolf, H. (2021). *Satellite Scheduling with VieSched++*, Master thesis, Technical University of Vienna.
- Schartner, M. & Böhm J. (2019). *VieSched++: A New VLBI Scheduling Software for Geodesy and Astrometry*, PASP 131(1002):084501, doi: <https://doi.org/10.1088/1538-3873/ab1820>
- VieSched++: <https://github.com/TUW-VieVS/VieSchedpp/>

# Appendix



## UT1 - UTC Dilution of Precision

$$A = \begin{pmatrix} \frac{\delta \Delta \tau_1}{\delta UT1} \\ \vdots \\ \frac{\delta \Delta \tau_n}{\delta UT1} \end{pmatrix}$$

$$N = A^T A = \left( \frac{\delta \Delta \tau_1}{\delta UT1} \right)^2 + \dots + \left( \frac{\delta \Delta \tau_n}{\delta UT1} \right)^2$$

- $A$  as design matrix
- $\Delta \tau_i$  as observed delay of reception times of two stations
- $\frac{\delta \Delta \tau_i}{\delta UT1}$  as partial derivative of  $\Delta \tau_i$  with respect to UT1

## UT1 - UTC Dilution of Precision

- UDOP factor as square root of trace of cofactor matrix  $Q_{xx}$
- sensitivity of VLBI observations to satellites to UT1-UTC (Belli, 2020)

$$Q_{xx} = N^{-1} = \frac{1}{\left(\frac{\delta\Delta\tau_1}{\delta UT1}\right)^2 + \dots + \left(\frac{\delta\Delta\tau_n}{\delta UT1}\right)^2}$$

$$UDOP = \sqrt{\frac{1}{\left(\frac{\delta\Delta\tau_1}{\delta UT1}\right)^2 + \dots + \left(\frac{\delta\Delta\tau_n}{\delta UT1}\right)^2}}$$

## UT1 - UTC Dilution of Precision (Belli, 2020)

- $a_k$  represents an entry in the design matrix for one baseline

$$a_k = \vec{e}_3 \cdot \left( \frac{\vec{y}}{R} \times \left( \frac{\vec{R}_1}{\rho_1} - \frac{\vec{R}_2}{\rho_2} \right) \right)$$

- $\vec{R}_1$  and  $\vec{R}_2$  as position vectors of the two VLBI antennas forming the baseline
- $\vec{y}$  as position vector of satellite in the Earth-fixed frame
- $\rho_i = \|\vec{y} - \vec{R}_i\|$  as range from each station to the satellite
- $R$  as Earth radius
- $\vec{e}_3$  as unit vector in z-direction
- $a_k$  represents the sensitivity of the range difference  $\Delta\rho$  to the Earth rotation angle
  - $\Delta\rho = \rho_1 - \rho_2$  as difference between the ranges from two stations, forming a baseline, to the satellite

## UT1 - UTC Dilution of Precision (Belli, 2020)

- UDOP factor as square root of trace of cofactor matrix  $Q_{xx}$

$$UDOP = \frac{1}{\sqrt{a_1^2 + a_2^2 + \dots + a_n^2}}$$

- $a_k$  with  $k = 1, 2, \dots, n$  represents entries in design matrix for  $n$  baselines
- low UDOP factor  $\rightarrow$  satellite observation is more sensitive to observe UT1-UTC

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